

EE 570: Location and Navigation

On-Line Bayesian Tracking

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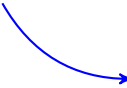
Sequentially estimate on-line the states of a system as it changes over time using observations that are corrupted with noise.

- *Filtering*: the time of the estimate coincides with the last measurement.
- *Smoothing*: the time of the estimate is within the span of the measurements.
- *Prediction*: the time of the estimate occurs after the last available measurement.

$$\vec{x}_k = \mathbf{f}_k(\vec{x}_{k-1}, \vec{w}_{k-1}) \quad (1)$$

$$\vec{z}_k = \mathbf{h}_k(\vec{x}_k, \vec{v}_k) \quad (2)$$

$(n \times 1)$ state vector at time k


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$$\vec{\mathbf{z}}_k = \mathbf{h}_k(\vec{\mathbf{x}}_k, \vec{\mathbf{v}}_k) \quad (2)$$


$(m \times 1)$ measurement vector at time k


Possibly non-linear function,

$$\mathbf{f}_k : \mathfrak{R}^n \times \mathfrak{R}^{n_w} \mapsto \mathfrak{R}^n$$

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i.i.d state noise

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The state process is Markov chain, i.e.,

$p(\vec{x}_k | \vec{x}_1, \dots, \vec{x}_{k-1}) = p(\vec{x}_k | \vec{x}_{k-1})$ and the distribution of \vec{z}_k conditional on the state \vec{x}_k is independent of previous state and measurement values, i.e., $p(\vec{z}_k | \vec{x}_{1:k}, \vec{z}_{1:k-1}) = p(\vec{z}_k | \vec{x}_k)$

Probabilistically estimate \vec{x}_k using previous measurement $\vec{z}_{1:k}$. In other words, construct the pdf $p(\vec{x}_k | \vec{z}_{1:k})$.

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Optimal MMSE Estimate

$$\mathbb{E}\{\|\vec{x}_k - \hat{\vec{x}}_k\|^2|\vec{z}_{1:k}\} = \int \|\vec{x}_k - \hat{\vec{x}}_k\|^2 p(\vec{x}_k|\vec{z}_{1:k}) d\vec{x}_k \quad (3)$$

in other words find the conditional mean

$$\hat{\vec{x}}_k = \mathbb{E}\{\vec{x}_k|\vec{z}_{1:k}\} = \int \vec{x}_k p(\vec{x}_k|\vec{z}_{1:k}) d\vec{x}_k \quad (4)$$

- \vec{w}_k and \vec{v}_k are drawn from a Gaussian distribution, uncorrelated have zero mean and statistically independent.

$$\mathbb{E}\{\vec{w}_k \vec{w}_i^T\} = \begin{cases} \mathbf{Q}_k & i = k \\ 0 & i \neq k \end{cases} \quad (5)$$

$$\mathbb{E}\{\vec{v}_k \vec{v}_i^T\} = \begin{cases} \mathbf{R}_k & i = k \\ 0 & i \neq k \end{cases} \quad (6)$$

$$\mathbb{E}\{\vec{w}_k \vec{v}_i^T\} = \begin{cases} 0 & \forall i, k \end{cases} \quad (7)$$

- \mathbf{f}_k and \mathbf{h}_k are both linear, e.g., the state-space system equations may be written as

$$\vec{\mathbf{x}}_k = \mathbf{\Phi}_{k-1} \vec{\mathbf{x}}_{k-1} + \vec{\mathbf{w}}_{k-1} \quad (8)$$

$$\vec{\mathbf{y}}_k = \mathbf{H}_k \vec{\mathbf{x}}_k + \vec{\mathbf{v}}_k \quad (9)$$

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($n \times n$) transition matrix relating $\vec{\mathbf{x}}_{k-1}$ to $\vec{\mathbf{x}}_k$

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$(m \times n)$ matrix provides noiseless connection between measurement and state vectors

$$\hat{\vec{x}}_{k|k-1} = \Phi_{k-1} \hat{\vec{x}}_{k-1|k-1} \quad (10)$$

$$\mathbf{P}_{k|k-1} = \mathbf{Q}_{k-1} + \Phi_{k-1} \mathbf{P}_{k-1|k-1} \Phi_{k-1}^T \quad (11)$$

$$\hat{\vec{x}}_{k|k} = \hat{\vec{x}}_{k|k-1} + \mathbf{K}_k (\vec{z}_k - \mathbf{H}_k \hat{\vec{x}}_{k|k-1}) \quad (12)$$

$$\mathbf{P}_{k|k} = (\mathbf{I} - \mathbf{K}_k \mathbf{H}_k) \mathbf{P}_{k|k-1} \quad (13)$$

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$(n \times m)$ Kalman gain

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Measurement innovation

$$\mathbf{K}_k = \mathbf{P}_{k|k-1} \mathbf{H}_k^T (\mathbf{H}_k \mathbf{P}_{k|k-1} \mathbf{H}_k^T + \mathbf{R}_k)^{-1} \quad (14)$$

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Covariance of the innovation term

Initial estimate ($\hat{\mathbf{x}}_0$ and \mathbf{P}_0)

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Compute Kalman gain

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Update estimate with measurement $\bar{\mathbf{z}}_k$

$$\hat{\mathbf{x}}_{k|k} = \hat{\mathbf{x}}_{k|k-1} + \mathbf{K}_k (\bar{\mathbf{z}}_k - \mathbf{H}_k \hat{\mathbf{x}}_{k|k-1})$$

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Update error covariance

$$\mathbf{P}_{k|k} = \mathbf{P}_{k|k-1} - \mathbf{K}_k \mathbf{H}_k \mathbf{P}_{k|k-1}$$

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$k = k + 1$

Project ahead

$$\hat{\mathbf{x}}_{k|k-1} = \Phi_{k-1} \hat{\mathbf{x}}_{k-1|k-1}$$

$$\mathbf{P}_{k|k-1} = \mathbf{Q}_{k-1} + \Phi_{k-1} \mathbf{P}_{k-1|k-1} \Phi_{k-1}^T$$

Initial estimate ($\hat{\mathbf{x}}_0$ and \mathbf{P}_0)

Compute Kalman gain

$$\mathbf{K}_k = \mathbf{P}_{k|k-1} \mathbf{H}_k^T (\mathbf{H}_k \mathbf{P}_{k|k-1} \mathbf{H}_k^T + \mathbf{R}_k)^{-1}$$

Project ahead

$$\begin{aligned} \hat{\mathbf{x}}_{k|k-1} &= \Phi_{k-1} \hat{\mathbf{x}}_{k-1|k-1} \\ \mathbf{P}_{k|k-1} &= \mathbf{Q}_{k-1} + \Phi_{k-1} \mathbf{P}_{k-1|k-1} \Phi_{k-1}^T \end{aligned}$$

Update estimate with measurement $\bar{\mathbf{z}}_k$

$$\hat{\mathbf{x}}_{k|k} = \hat{\mathbf{x}}_{k|k-1} + \mathbf{K}_k (\bar{\mathbf{z}}_k - \mathbf{H}_k \hat{\mathbf{x}}_{k|k-1})$$

Update error covariance

$$\mathbf{P}_{k|k} = \mathbf{P}_{k|k-1} - \mathbf{K}_k \mathbf{H}_k \mathbf{P}_{k|k-1}$$

 $k = k + 1$

If R is a block matrix, i.e., $R = \text{diag}(R^1, R^2, \dots, R^r)$. The R^i has dimensions $p^i \times p^i$. Then, we can sequentially process the measurements as:

For $i = 1, 2, \dots, r$

$$\mathbf{K}^i = \mathbf{P}^{i-1}(\mathbf{H}^i)^T (\mathbf{H}^i \mathbf{P}^{i-1} (\mathbf{H}^i)^T + \mathbf{R}^i)^{-1} \quad (15)$$

$$\hat{\mathbf{x}}_{k|k}^i = \hat{\mathbf{x}}_{k|k}^i + \mathbf{K}^i (\mathbf{z}_k^i - \mathbf{H}^i \hat{\mathbf{x}}_{k|k}^{i-1}) \quad (16)$$

$$\mathbf{P}^i = (\mathbf{I} - \mathbf{K}^i \mathbf{H}^i) \mathbf{P}^{i-1} \quad (17)$$

where $\hat{\mathbf{x}}_{k|k}^0 = \hat{\mathbf{x}}_{k|k-1}$, $\mathbf{P}^0 = \mathbf{P}_{k|k-1}^0$ and \mathbf{H}^i is $p^i \times n$ corresponding to the rows of \mathbf{H} corresponding the measurement being processed.

The system is observable if the observability matrix

$$\mathcal{O}(k) = \begin{bmatrix} \mathbf{H}(k - n + 1) \\ \mathbf{H}(k - n - 2)\Phi(k - n + 1) \\ \vdots \\ \mathbf{H}(k)\Phi(k - 1) \dots \Phi(k - n + 1) \end{bmatrix} \quad (18)$$

where n is the number of states, has a rank of n . The rank of \mathcal{O} is a binary indicator and does **not** provide a measure of how close the system is to being unobservable, hence, is prone to numerical ill-conditioning.

In addition to the computation of the rank of $\mathcal{O}(k)$, compute the Singular Value Decomposition (SVD) of $\mathcal{O}(k)$ as

$$\mathcal{O} = U\Sigma V^* \quad (19)$$

and observe the diagonal values of the matrix Σ . Using this approach it is possible to monitor the variations in the system observability due to changes in system dynamics.

- Kalman filter is optimal under the aforementioned assumptions,
- and it is also an unbiased and minimum variance estimate.
- If the Gaussian assumptions is not true, Kalman filter is biased and not minimum variance.
- Observability is dynamics dependent.
- The error covariance update may be implemented using the *Joseph form* which provides a more stable solution due to the guaranteed symmetry.

$$P_{k|k} = (I - K_k H_k) P_{k|k-1} (I - K_k H_k)^T + K_k R_k K_k^T \quad (20)$$

$$\dot{\vec{x}}(t) = \mathbf{F}(t)\vec{x}(t) + \mathbf{G}(t)\vec{w}(t) \quad (21)$$

To obtain the state vector estimate $\hat{\vec{x}}(t)$

$$\mathbb{E}\{\dot{\vec{x}}(t)\} = \frac{\partial}{\partial t} \hat{\vec{x}}(t) = \mathbf{F}(t)\hat{\vec{x}}(t) \quad (22)$$

Solving the above equation over the interval $t - \tau_s, t$

$$\hat{\vec{x}}(t) = e^{(\int_{t-\tau_s}^t \mathbf{F}(t') dt')} \hat{\vec{x}}(t - \tau_s) \quad (23)$$

where \mathbf{F}_{k-1} is the average of \mathbf{F} at times t and $t - \tau_s$.

As shown in the Kalman filter equations the state vector estimate is given by

$$\hat{\mathbf{x}}_{k|k-1} = \mathbf{\Phi}_{k-1} \hat{\mathbf{x}}_{k-1|k-1}$$

Therefore,

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Therefore,

$$\mathbf{\Phi}_{k-1} = e^{\mathbf{F}_{k-1}\tau_s} \approx \mathbf{I} + \mathbf{F}_{k-1}\tau_s \quad (24)$$

where \mathbf{F}_{k-1} is the average of \mathbf{F} at times t and $t - \tau_s$, and first order approximation is used.

Assuming white noise, small time step, \mathbf{G} is constant over the integration period, and the trapezoidal integration

$$\mathbf{Q}_{k-1} \approx \frac{1}{2} \left[\Phi_{k-1} \mathbf{G}_{k-1} \mathbf{Q}(t_{k-1}) \mathbf{G}_{k-1}^T \Phi_{k-1}^T + \mathbf{G}_{k-1} \mathbf{Q}(t_{k-1}) \mathbf{G}_{k-1}^T \right] \tau_s \quad (25)$$

where

$$\mathbb{E}\{\vec{\mathbf{w}}(\eta) \vec{\mathbf{w}}^T(\zeta)\} = \mathbf{Q}(\eta) \delta(\eta - \zeta) \quad (26)$$

$$\mathbf{F}_k = \left. \frac{\partial \mathbf{f}(\vec{x})}{\partial \vec{x}} \right|_{\vec{x} = \hat{\vec{x}}_{k|k-1}}, \quad \mathbf{H}_k = \left. \frac{\partial \mathbf{h}(\vec{x})}{\partial \vec{x}} \right|_{\vec{x} = \hat{\vec{x}}_{k|k-1}} \quad (27)$$

where

$$\frac{\partial \mathbf{f}(\vec{x})}{\partial \vec{x}} = \begin{pmatrix} \frac{\partial f_1}{\partial x_1} & \frac{\partial f_1}{\partial x_2} & \cdots \\ \frac{\partial f_2}{\partial x_1} & \frac{\partial f_2}{\partial x_2} & \cdots \\ \vdots & \ddots & \vdots \end{pmatrix}, \quad \frac{\partial \mathbf{h}(\vec{x})}{\partial \vec{x}} = \begin{pmatrix} \frac{\partial h_1}{\partial x_1} & \frac{\partial h_1}{\partial x_2} & \cdots \\ \frac{\partial h_2}{\partial x_1} & \frac{\partial h_2}{\partial x_2} & \cdots \\ \vdots & \ddots & \vdots \end{pmatrix} \quad (28)$$

State Equation

$$\dot{b}(t) = -\frac{1}{T_c}b(t) + w(t) \quad (29)$$

Autocorrelation Function

$$\mathbb{E}\{b(t)b(t + \tau)\} = \sigma_{BI}^2 e^{-|\tau|/T_c} \quad (30)$$

where

$$\mathbb{E}\{w(t)w(t + \tau)\} = Q(t)\delta(t - \tau) \quad (31)$$

$$Q(t) = \frac{2\sigma_{BI}^2}{T_c} \quad (32)$$

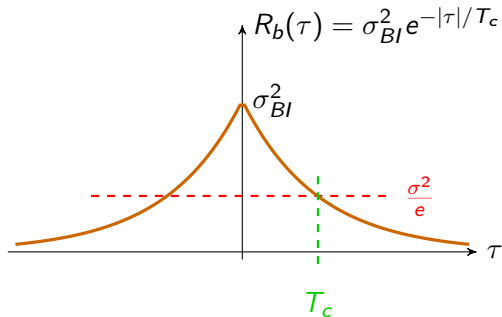
and T_c is the correlation time.

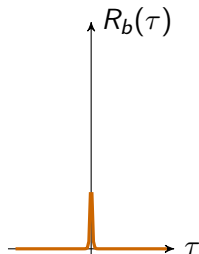
State Equation

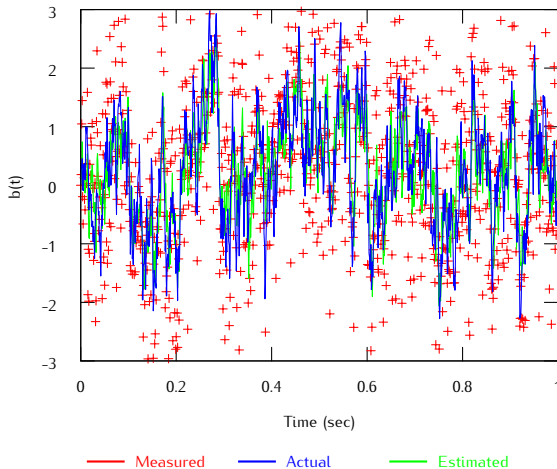
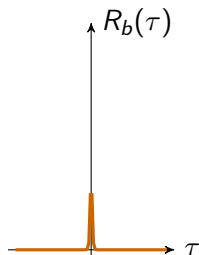
$$b_k = e^{-\frac{1}{T_c} T_s} b_{k-1} + w_{k-1} \quad (33)$$

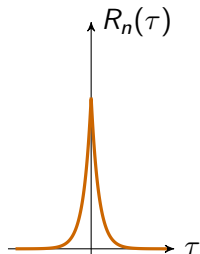
System Covariance Matrix

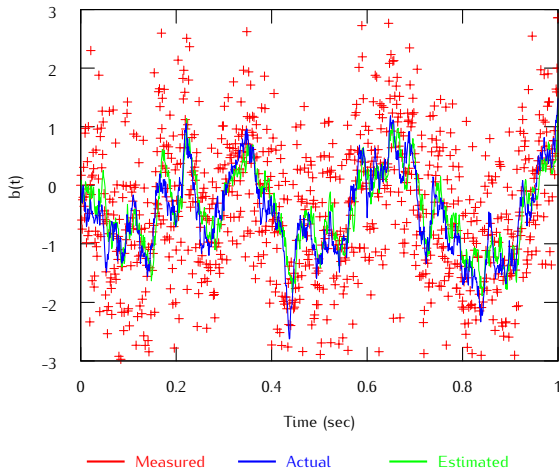
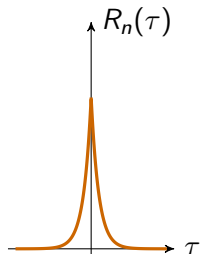
$$Q = \sigma_{BI}^2 [1 - e^{-\frac{2}{T_c} T_s}] \quad (34)$$











Propagates carefully chosen sample points (using unscented transformation) through the true non-linear system, and therefore captures the posterior mean and covariance accurately to the second order.

A Monte Carlo based method. It allows for a complete representation of the state distribution function. Unlike EKF and UKF, particle filters do not require the Gaussian assumptions.

Bayesian Filtering: From Kalman Filters to Particle Filters, and Beyond

by Zhe Chen