

Lecture

Navigation Equations: ECEF Mechanization

EE 570: Location and Navigation

Lecture Notes Update on February 20, 2014

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ECEF Mechanization

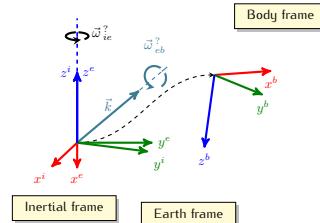
- Determine the position, velocity and attitude of the **body** frame *wrt* the **Earth** frame.
 - **Position** — Vector from the origin of the earth frame to the origin of the body frame resolved in the earth frame: \vec{r}_{eb}^e
 - **Velocity** — Velocity of the body frame *wrt* the earth frame resolved in the earth frame: \vec{v}_{eb}^e
 - **Attitude** — Orientation of the body frame *wrt* the earth frame: C_b^e

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Attitude — Method A

- Body orientation frame at time “ k ” *wrt* time “ $k - 1$ ”
 - $\Delta t = t_k - t_{k-1}$
- Start with angular velocity

$$\begin{aligned}\vec{\omega}_{ib}^e &= \vec{\omega}_{ie}^e + \vec{\omega}_{eb}^e \\ \vec{\omega}_{eb}^e &= C_b^e \vec{\omega}_{ib}^b - \vec{\omega}_{ie}^e \\ \Omega_{eb}^e &= C_b^e \Omega_{ib}^b C_e^b - \Omega_{ie}^e \\ C_b^e(+) - C_b^e(-) &\approx \Delta t \Omega_{eb}^e C_b^e(-) \\ C_b^e(+) &\approx C_b^e(-) + \Delta t (C_b^e \Omega_{ib}^b C_e^b - \Omega_{ie}^e) C_b^e(-) \\ &= C_b^e(-) (\mathcal{I} + \Omega_{ib}^b \Delta t) - \Omega_{ie}^e C_b^e(-) \Delta t\end{aligned}$$



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Attitude — Method B

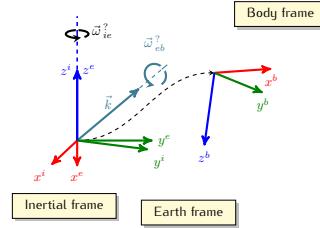
- Body orientation frame at time “ k ” *wrt* time “ $k - 1$ ”
 - $\Delta t = t_k - t_{k-1}$
- Start with the angular velocity

$$\Omega_{eb}^e = C_b^e \Omega_{ib}^b C_e^b - \Omega_{ie}^e$$

$$C_b^e(+) = C_b^e(-) e^{\Omega_{eb}^b \Delta t} = e^{\Omega_{eb}^e \Delta t} C_b^e(-)$$

$$C_b^e(+) = [\mathcal{I} + \sin(\Delta\theta)\mathbf{\tilde{\kappa}} + [1 - \cos(\Delta\theta)]\mathbf{\tilde{\kappa}}^2] C_b^e(-)$$

$$e^{\Omega_{eb}^e \Delta t} = e^{\tilde{\kappa}\theta}$$



Attitude — Method C

- Body orientation frame at time "k" wrt time "k - 1"

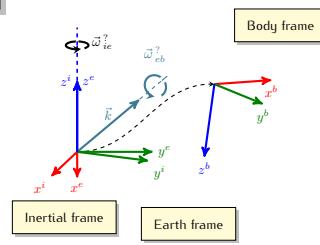
$$-\Delta t = t_k - t_{k-1}$$

$$\vec{\omega}_{eb}^e \Delta t = \vec{k} \Delta \theta$$

$$\bar{q}_b^e(+) = \Delta \bar{q}_b^e \otimes \bar{q}_b^e(-)$$

$$\Delta \bar{q}_b^e = \begin{bmatrix} \cos(\frac{\Delta\theta}{2}) \\ \vec{k} \sin(\frac{\Delta\theta}{2}) \end{bmatrix}$$

Need to periodically renormalize \bar{q}



Attitude Update— Summary

- High fidelity

$$C_b^e(+) = [\mathcal{I} + \sin(\Delta\theta)\mathbf{\tilde{\kappa}} + [1 - \cos(\Delta\theta)]\mathbf{\tilde{\kappa}}^2] C_b^e(-) \quad (1)$$

or

$$\bar{q}_b^e(+) = \begin{bmatrix} \cos(\frac{\Delta\theta}{2}) \\ \vec{k} \sin(\frac{\Delta\theta}{2}) \end{bmatrix} \otimes \bar{q}_b^e(-) \quad (2)$$

- Low fidelity

$$C_b^e(+) \approx C_b^e(-) (\mathcal{I} + \Omega_{ib}^b \Delta t) - \Omega_{ie}^e C_b^e(-) \Delta t \quad (3)$$

Steps 2–4

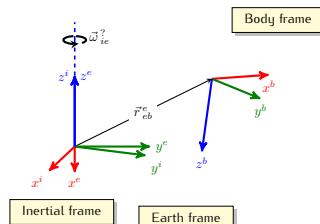
2. Specific force transformation

- Simply coordinatize the specific force

$$\vec{f}_{ib}^e = C_b^e(+) \vec{f}_{ib}^b \quad (4)$$

3. Velocity update

$$\begin{aligned} \vec{r}_{ib}^i &= \cancel{\vec{r}_{ie}^i}^0 + C_i^e \vec{r}_{eb}^e \Rightarrow \vec{r}_{eb}^e = C_i^e \vec{r}_{ib}^i \\ \vec{v}_{eb}^e &= \dot{\vec{r}}_{eb}^e \\ &= \dot{C}_i^e \vec{r}_{ib}^i + C_i^e \dot{\vec{r}}_{ib}^i \\ &= \Omega_{ei}^e C_i^e \vec{r}_{ib}^i + C_i^e \vec{v}_{ib}^i \\ &= -\Omega_{ie}^e \vec{r}_{eb}^e + C_i^e \vec{v}_{ib}^i \end{aligned}$$



Steps 2–4

$$\begin{aligned}
C_i^e \vec{v}_{ib}^i &= \Omega_{ie}^e \vec{r}_{eb}^e + \vec{v}_{eb}^e \\
\vec{a}_{eb}^e &= \dot{\vec{v}}_{eb}^e = \frac{d}{dt} (-\Omega_{ie}^e \vec{r}_{eb}^e + C_i^e \vec{v}_{ib}^i) \\
&= -\Omega_{ie}^e \dot{\vec{r}}_{eb}^e + \dot{C}_i^e \vec{v}_{ib}^i + C_i^e \dot{\vec{v}}_{ib}^i \\
&= -\Omega_{ie}^e \vec{v}_{eb}^e + \Omega_{ie}^e C_i^e \vec{v}_{ib}^i + C_i^e \vec{a}_{ib}^i \\
&= -\Omega_{ie}^e \vec{v}_{eb}^e - \Omega_{ie}^e [\vec{v}_{eb}^e + \Omega_{ie}^e \vec{r}_{eb}^e] + C_i^e \vec{a}_{ib}^i \\
&= -2\Omega_{ie}^e \vec{v}_{eb}^e - \Omega_{ie}^e \Omega_{ie}^e \vec{r}_{eb}^e + \vec{a}_{ib}^e \\
&= -2\Omega_{ie}^e \vec{v}_{eb}^e + \vec{f}_{ib}^e + \vec{g}_b^e \\
\vec{v}_{eb}^e(+) &= \vec{v}_{eb}^e(-) + \vec{a}_{eb}^e \Delta t \\
&= \vec{v}_{eb}^e(-) + [\vec{f}_{ib}^e + \vec{g}_b^e - 2\Omega_{ie}^e \vec{v}_{eb}^e(-)] \Delta t \quad (5)
\end{aligned}$$

Steps 2–4

4. Position update

- by simple numerical integration

$$\vec{r}_{eb}^e(+) = \vec{r}_{eb}^e(-) + \vec{v}_{eb}^e(-) \Delta t + \vec{a}_{eb}^e \frac{\Delta t^2}{2} \quad (6)$$

ECEF Mechanization Summary

$$C_b^e(+) = [\mathcal{I} + \sin(\Delta\theta)\hat{\mathbf{k}} + [1 - \cos(\Delta\theta)]\hat{\mathbf{k}}^2] C_b^e(-)$$

or

$$C_b^e(+) \approx C_b^e(-) (\mathcal{I} + \Omega_{ib}^b \Delta t) - \Omega_{ie}^i C_b^e(-) \Delta t$$

or

$$\vec{q}_b^e(+) = \begin{bmatrix} \cos(\frac{\Delta\theta}{2}) \\ \vec{k} \sin(\frac{\Delta\theta}{2}) \end{bmatrix} \otimes \vec{q}_b^e(-)$$

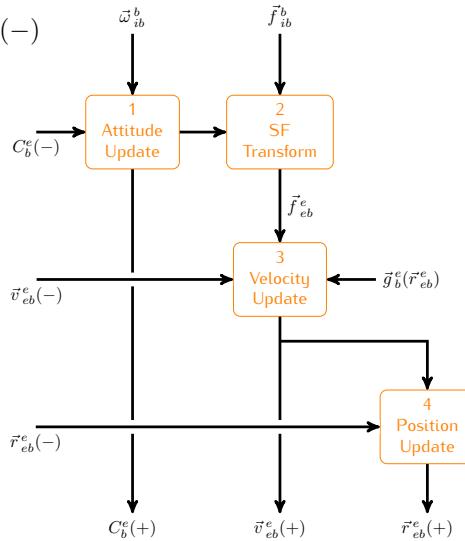
and

$$\vec{f}_{ib}^e = C_b^e(+) \vec{f}_{ib}^b$$

$$\vec{a}_{eb}^e = \vec{f}_{ib}^e + \vec{g}_b^e - 2\Omega_{ie}^i \vec{v}_{eb}^e(-)$$

$$\vec{v}_{eb}^e(+) = \vec{v}_{eb}^e(-) + \vec{a}_{eb}^e \Delta t$$

$$\vec{r}_{eb}^e(+) = \vec{r}_{eb}^e(-) + \vec{v}_{eb}^e(-) \Delta t + \vec{a}_{eb}^e \frac{\Delta t^2}{2}$$



ECEF Mechanization — Continuous Case

In continuous time notation

- Attitude: $\dot{C}_b^e = C_b^e \Omega_{eb}^b$ or $\dot{\vec{q}}_b^e = \frac{1}{2} [\check{\omega}_{eb}^b \circledast] \vec{q}_b^e(t)$
- Velocity: $\dot{\vec{v}}_{eb}^e = C_b^e \vec{f}_{ib}^b + \vec{g}_b^e - 2\Omega_{ie}^i \vec{v}_{eb}^e$
- Position: $\dot{\vec{r}}_{ib}^e = \vec{v}_{eb}^e$

- In State-space notation

$$\begin{bmatrix} \dot{\vec{r}}_{eb}^e \\ \dot{\vec{v}}_{eb}^e \\ \dot{C}_b^i \end{bmatrix} = \begin{bmatrix} \vec{v}_{eb}^e \\ C_b^e \vec{f}_{ib}^b + \vec{g}_b^e - 2\Omega_{ie}^i \vec{v}_{eb}^e \\ C_b^e \Omega_{eb}^b \end{bmatrix} \quad (7)$$

or

$$\begin{bmatrix} \dot{\vec{r}}_{eb}^e \\ \dot{\vec{v}}_{eb}^e \\ \dot{\vec{q}}_b^e \end{bmatrix} = \begin{bmatrix} \vec{v}_{eb}^e \\ C_b^e \vec{f}_{ib}^b + \vec{g}_b^e - 2\Omega_{ie}^i \vec{v}_{eb}^e \\ \frac{1}{2} [\check{\omega}_{eb}^b \otimes] \bar{q}_b^e(t) \end{bmatrix} \quad (8)$$

where $\vec{\omega}_{ib}^b = \vec{\omega}_{ie}^b + \vec{\omega}_{eb}^b$, and $\Omega_{eb}^b = \Omega_{ib}^b - \Omega_{ie}^b$.

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Appendix

$$[\bar{q} \otimes] = \begin{bmatrix} q_s & -q_x & -q_y & -q_z \\ q_x & q_s & -q_z & q_y \\ q_y & q_z & q_s & -q_x \\ q_z & -q_y & q_x & q_s \end{bmatrix}$$

$$[\bar{q} \otimes] = \begin{bmatrix} q_s & -q_x & -q_y & -q_z \\ q_x & q_s & q_z & -q_y \\ q_y & -q_z & q_s & q_x \\ q_z & q_y & -q_x & q_s \end{bmatrix}$$

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