

1. Given the geodetic coordinates of the peak of Mt. Everest as Latitude (L_b) 27deg 59min 16sec N, Longitude (λ_b) 86deg 56min 40sec E, and height (h_b) 8850 meters (derived by GPS in 1999):

- (a) Develop a MATLAB function

```
function [r_e__e_b]=llh2xyz(L_b,lambda_b,h_b)
```

to convert from geodetic curvilinear lat, lon and height to ECEF rectangular x , y and z coordinates (use SI units).

- i. Attach a printout of your function or put a copy in the shared folder.
- ii. Test your llh2xyz function using coordinates of the peak of Mt Everest. What is \vec{r}_{eb}^e at the peak?

- (b) Develop a MATLAB function

```
function [L_b,lambda_b,h_b] = xyz2llh(r_e__e_b)
```

to convert ECEF x , y , and z coordinates to lat, lon, and height (use SI units). HINT: This should be an iterative transformation (i.e., not closed form).

- i. Attach a printout of your function or put a copy in the shared folder.
- ii. Test your xyz2llh function with the ECEF coords obtained from part (1a).

- (c) Develop a MATLAB function

```
function [C_e__n] = llh2dcm(L_b, lambda_b, h_b)
```

to compute the orientation (as a rotation/cosine matrix) of the navigation frame relative to the ECEF frame given geodetic lat and lon.

- i. Attach a printout of your function or put a copy in the shared folder.
- ii. Use your function to obtain the orientation of the navigation frame at the peak of Mt Everest relative to the ECEF frame.
- iii. Given a body is on the peak of Mt Everest facing east, obtain the orientation of the body relative to the ECEF frame.

- (d) What is the acceleration due to gravity at the ellipsoid (i.e., at the ellipsoid $h_b = 0$). HINT: This should only be a function of lat—see page 70 of text)?

- (e) What is the magnitude of the centrifugal acceleration ($-\Omega_{ie}^e \Omega_{ie}^e \vec{r}_{eb}^e$) at the ellipsoid and at the peak?

- (f) What is the magnitude of the gravitational attraction at the ellipsoid and at the peak? HINT: See page 72 to compute $\vec{\gamma}_{ib}^e = \vec{\gamma}_{ib}^i |_{\vec{r}_{ib}^i = \vec{r}_{eb}^e}$.

2. Develop a MATLAB function

```
function [g_n_bD] = gravity(L_b,h_b)
```

(see eqn. 2.139 on page 71 of Groves) to approximate the “down” component of the acceleration due to gravity as a function of lat & height (Please use SI units).

- (a) Use this function to compute the acceleration due to gravity at the peak of Mt Everest.
- (b) What is the difference in m/s^2 between the answer obtained in questions 2(a) and that of 1(d)?
 - i. Based on this difference how much less would you weigh at the peak than at the ellipsoid (in lbs)?

3. Develop the following MATLAB functions

- (a) `function [q] = dcm2q(C)`
- (b) `function [C] = q2dcm(q)`
- (c) `function [q] = q1xq2(q1,q2)`
for \otimes
- (d) `function [q] = q1starq2(q1,q2)`
for \otimes .

Use some test cases to check they function properly, and turn in a printout of the functions or place a copy of them in the shared folder.

4. Given the quaternion $\bar{\mathbf{q}}_{ab}^a(t)$ describes rotation about a z -axis by time-varying angle $\theta(t)$, use the relationship $\dot{\bar{\mathbf{q}}}_b^a(t) = \frac{1}{2}[\check{\omega}_{ab}^a \otimes] \bar{\mathbf{q}}_b^a(t)$ derived in class to find the corresponding angular velocity $\vec{\omega}$.